



Technical Standard Order

Subject: AIRBORNE NAVIGATION SENSORS USING THE GLOBAL POSITIONING SYSTEM (GPS) AUGMENTED BY THE WIDE AREA AUGMENTATION SYSTEM (WAAS)

1. PURPOSE. This Technical Standard Order (TSO) tells persons seeking a TSO authorization or letter of design approval what minimum performance standards (MPS) their airborne navigation sensors, using the Global Positioning System (GPS) augmented by the Wide Area Augmentation System (WAAS), must first meet in order to obtain approval and be identified with the applicable TSO marking.

2. APPLICABILITY. This TSO is effective for new applications submitted after the effective date of this TSO. All prior revisions to this TSO are no longer effective and applications will not be accepted after the effective date of this TSO.

3. REQUIREMENTS. New models of Airborne navigation sensors using GPS augmented by WAAS that are to be so identified and that are manufactured on or after the effective date of this TSO must meet the MPS for Class Beta equipment in Section 2 of RTCA/DO-229C, "Minimum Operational Performance Standards for Global Positioning System/Wide Area Augmentation System Airborne Equipment," dated November 28, 2001. Class Beta equipment is defined in Section 1 of RTCA/DO-229C.

a. Functionality. The standards of this TSO apply to equipment intended to provide position information to a navigation management unit that outputs deviation commands referenced to a desired flight path. These deviations will be used by the pilot or autopilot to guide the aircraft. These standards do not address integration issues with other avionics, such as the potential for the sensor to inadvertently command an autopilot hardover. These standards also do not address the use of position information for other applications such as automatic dependent surveillance.

b. Failure Condition Classification. Failure of the function defined in paragraph 3 and 3a of this TSO has been determined to be: a major failure condition for loss of function and malfunction of en route, terminal, or nonprecision approach position data; a major failure condition for loss of function of precision approach position data; and a hazardous failure condition for the malfunction of precision approach position data. The applicant must develop the system to at least the design assurance level commensurate with this hazard classification.

c. Functional Qualification. The required performance shall be demonstrated under the test conditions and procedures specified in RTCA/DO-229C, Section 2.5. The use of test procedures

other than those specified in Sections 2.5.2 through 2.5.9 of RTCA/DO-229C constitutes a deviation to this TSO.

d. Environmental Qualifications. The equipment shall be subject to the test conditions as specified in RTCA/DO-160D, "Environmental Conditions and Test Procedures for Airborne Equipment," dated July 29, 1997.

e. Software Qualifications. If the article includes software, the software must be developed in accordance with Sections 3-11 and Annex A of RTCA/DO-178B, "Software Considerations in Airborne Systems and Equipment Certification," dated December 1, 1992.

f. Barometric-Aided Fault Detection and Exclusion (FDE). If the equipment uses barometric-aiding to enhance the availability of FDE, it must meet the requirements in Appendix G of RTCA/DO-229C.

g. Deviations. The FAA has provisions for using alternative or equivalent means of compliance to the criteria set forth in the MPS of this TSO. Applicants invoking these provisions shall demonstrate that an equivalent level of safety is maintained and shall apply for a deviation per 14 CFR § 21.609.

4. MARKING. Under 14 CFR § 21.607(d), articles manufactured under this TSO must be marked as follows:

a. At least one major component must be permanently and legibly marked with all of the information listed in 14 CFR § 21.607(d), except for the following: the option in 14 CFR § 21.607(d)(2), where the name, type and part number must be used in lieu of the optional model number; and the option in 14 CFR § 21.607(d)(3), where the date of manufacture must be used in lieu of the optional serial number.

b. In addition to the requirements of 14 CFR § 21.607(d), each separate component that is easily removable (without hand tools), each interchangeable element, and each separate sub-assembly of the article that the manufacturer determines may be interchangeable must be permanently and legibly marked with at least the name of the manufacturer, manufacturer's subassembly part number, and the TSO number.

c. If the component includes software, the part number must include hardware and software identification, or a separate part number may be utilized for hardware and software. Either approach must include a means for showing the modification status. Note that similar software versions, which have been approved to different software levels, must be differentiated by part number.

d. At least one major component must be permanently and legibly marked with the operational equipment class as defined in Section 1 of RTCA/DO-229C (e.g., Class 2).

e. When applicable, identification that the article is an incomplete system or that the article accomplishes additional functions beyond that described in paragraph 3 and 3a of this TSO.

5. DATA REQUIREMENTS.

a. **Application Data.** Under 14 CFR § 21.605(a)(2), the manufacturer must furnish the Manager, Aircraft Certification Office (ACO), Federal Aviation Administration (FAA), responsible for the manufacturer's facilities, one copy each of the following technical data to support the FAA design and production approval:

(1) Operating instructions and equipment limitations. The limitations shall be sufficient to describe the operational capability of the equipment.

(2) Installation procedures and limitations. The limitations shall be sufficient to ensure that the article, when installed in accordance with the installation procedures, continues to meet the requirements of this TSO. The limitations shall identify any unique aspects of the installation. The limitations shall include at least the following:

(a) A note with the following statement:

The conditions and tests required for TSO approval of this article are minimum performance standards. It is the responsibility of those installing this article either on or within a specific type or class of aircraft to determine that the aircraft installation conditions are within the TSO standards. TSO articles must have separate approval for installation in an aircraft. The article may be installed only if performed under 14 CFR part 43 or the applicable airworthiness requirements.

(b) Adequate specification of the interface between the GPS/WAAS sensor and other systems to ensure proper functioning of the integrated system. This must include maximum tolerable currents and voltages into the antenna port if the equipment is to be installed with a standard antenna (TSO-C144).

(c) If the equipment has only been demonstrated to satisfy the requirements of RTCA/DO-229C when used in conjunction with a particular antenna, the use of that antenna (by part number) must be specified as a limitation.

(d) If the equipment is dependent on any inputs in order to satisfy the requirements of RTCA/DO-229C (e.g., baro-aided FDE), those inputs should be made a requirement on the installation (i.e., a limitation).

(e) If the software qualification limits the equipment to be eligible on certain aircraft types, identification of the qualification level and identification that the equipment has not been determined to be eligible for all aircraft types (e.g., AC 23-1309-1A states that the DO-178B Level C software may be associated with a hazardous failure condition for certain aircraft types).

(f) When applicable, identification that the article is an incomplete system or a multi-use system. This must describe the functions that are intended to be provided by the article.

(3) Schematic drawings, as applicable to the installation procedures.

(4) Wiring drawings, as applicable to the installation procedures.

(5) List of the components, by part number, that make up the equipment complying with the standards prescribed in this TSO. Manufacturers should include vendor part number cross-references when applicable.

(6) Instructions, in the form of a Component Maintenance Manual (CMM) containing information on the periodic maintenance, calibration and repair, for the continued airworthiness of installed GPS/WAAS airborne equipment, including recommended inspection intervals and service life.

(7) An environmental qualifications form as described in RTCA/DO-160D for each component of the system.

(8) Materials and process specifications list.

(9) The quality control system description required by 14 CFR §§ 21.605(a)(3) and 21.143(a) including functional test specifications to be used to test each production article to ensure compliance with this TSO.

(10) Manufacturer's TSO qualification test report.

(11) Nameplate drawing providing the information required by paragraph 4 of this TSO.

(12) A list of all drawings and processes, including revision level, necessary to define the article's design. In the case of a minor change, any revisions to the drawing list need only be made available upon request.

(13) If the article includes software: Plan for Software Aspects of Certification (PSAC); Software Configuration Index; and Software Accomplishment Summary. The FAA recommends that the PSAC be submitted early in the software development process. Early submittal will allow timely resolution of issues such as partitioning and determination of software levels.

b. Manufacturer Data. In addition to the data to be furnished directly to the FAA, each manufacturer must have available for review by the manager of the ACO responsible for the manufacturer's facilities the following technical data:

(1) The functional qualification specifications to be used to qualify each production article to ensure compliance with this TSO.

- (2) Equipment calibration procedures.
- (3) Corrective maintenance procedures within 12 months after TSO authorization.
- (4) Schematic drawings.
- (5) Wiring diagrams.
- (6) Material and process specifications.
- (7) The results of the environmental qualification tests conducted in accordance with RTCA/DO-160D.
- (8) If the article includes software, the appropriate documentation as defined in RTCA/DO-178B, including all data supporting the applicable objectives found in Annex A of RTCA/DO-178B, Process Objectives and Outputs by Software Level.
- (9) Qualification test procedures used to determine compliance with this TSO.

c. Furnished Data.

(1) One copy of the data and information specified in paragraphs 5(a)(1) through (7) of this TSO and any other data or information necessary for the proper installation, certification and use and/or for continued airworthiness of the GPS/WAAS airborne equipment, must accompany each article manufactured under this TSO.

(2) If the appliance accomplishes any additional functions beyond that described in paragraphs 3 and 3a of this TSO, then a copy of the data and information specified in paragraphs 5a(12) and (13) must also go to each person receiving for use one or more articles manufactured under this TSO.

6. AVAILABILITY OF REFERENCED DOCUMENTS.

a. Copies of RTCA/DO-160D, DO-178B, and DO-229C may be purchased from RTCA Inc., 1828 L Street, NW, Suite 805, Washington, D.C. 20036, or their website @ www.rtca.org.

b. You may buy copies of Federal Aviation Regulations 14 CFR part 21, Subpart O, from the Superintendent of Documents, Government Printing Office, Washington, D.C. 20402-9325. Copies also can be obtained from the Government Printing Office (GPO), electronic CFR Internet website @ www.access.gpo.gov/ecfr/.

c. Advisory Circular (AC) 20-110 *{or current revision}*, "Index of Aviation Technical Standard Orders," and AC 20-36 *{or current revision}*, "Index of Articles Certified under the Technical Standard Order System" *{list other applicable or referenced ACs, as needed}* may be

obtained from the U.S. Department of Transportation, Subsequent Distribution Office, Ardmore East Business Center, 3341 Q 75th Avenue, Landover, MD 20785, telephone (301) 322-4477 or FAX (301) 386-5394. Copies also may be obtained from the FAA Internet website @ www.faa.gov/avr/air/airhome.htm and select from the “Available Information” drop down list.

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